

# Cosmology and astrophysics with Planck

## Satellite, instruments and data processing

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# Particular thanks for some of the material to:

- ▶ Fabrizio Villa and Maura Sandri (INAF-IASF Bologna) for material on optics
- ▶ Gianluca Morgante (INAF-IASF Bologna) and JPL team for material on Sorption Cooler
- ▶ ESA for material on orbit and scanning strategy



# Outline

Overall description of Planck

From the scientific goals to technology requirements

Orbit and scanning strategy

Optics

- Telescope

- Feed horns

Instruments

- High Frequency Instrument

- Low Frequency Instrument

Thermal design

- Passive thermal design

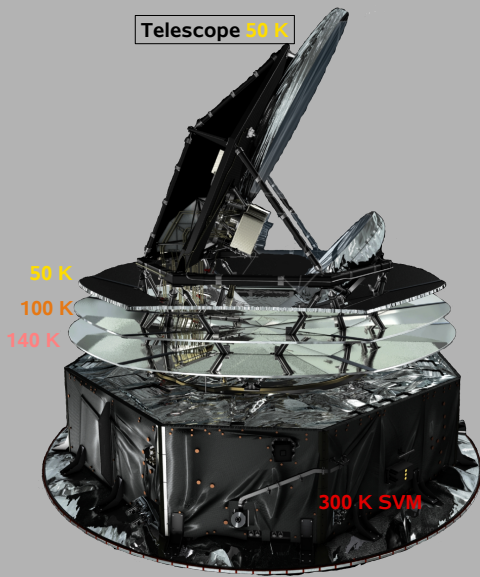
- Active cooling

- The Planck sorption cooler

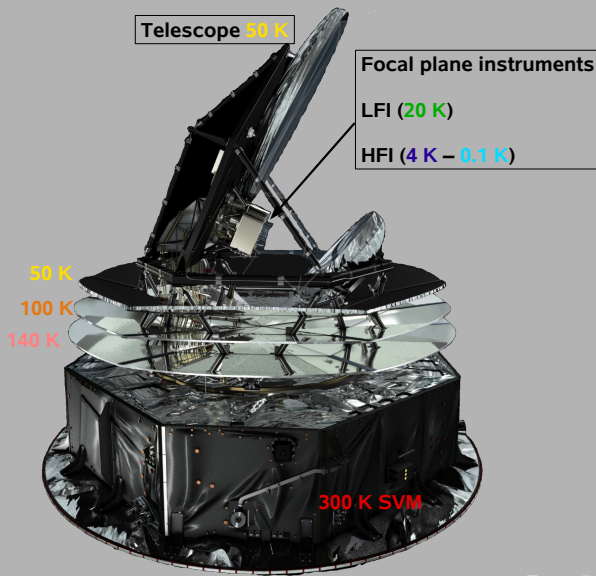
Data processing



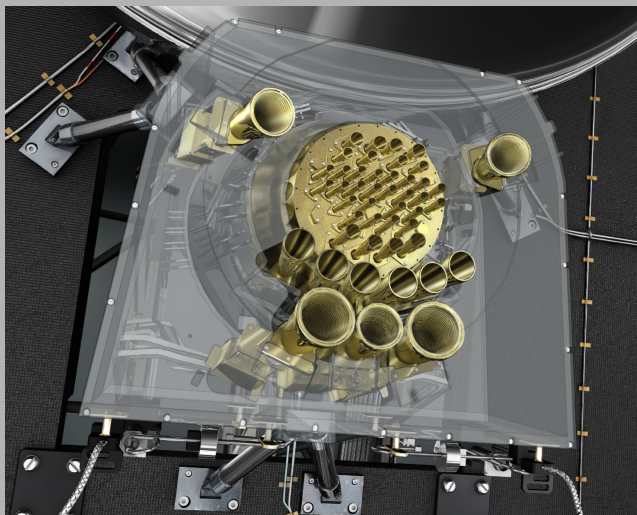
# The Planck satellite



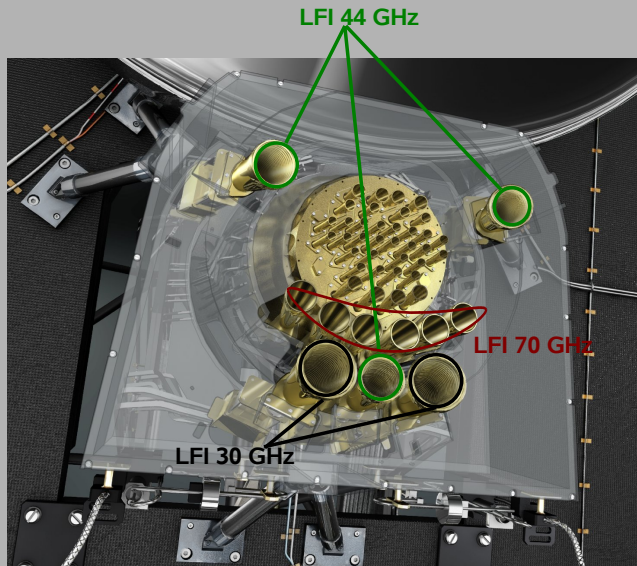
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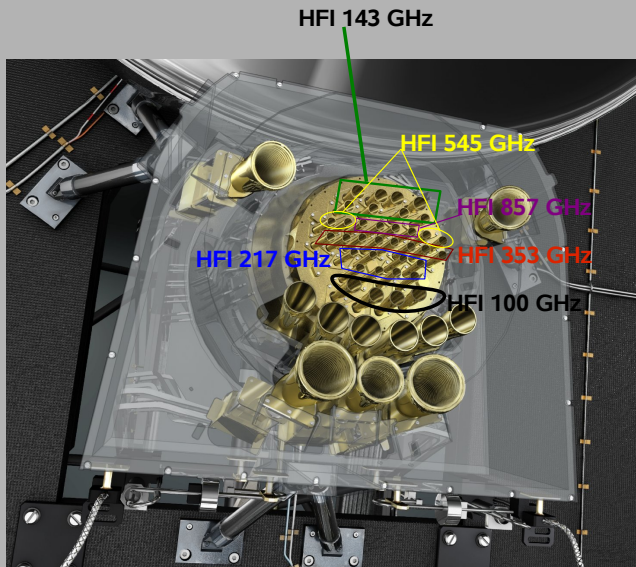
# Planck instruments



# Planck instruments - LFI



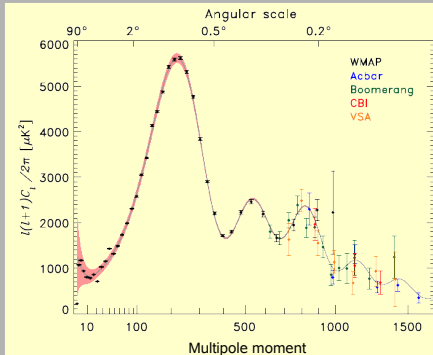
## Planck instruments - HFI



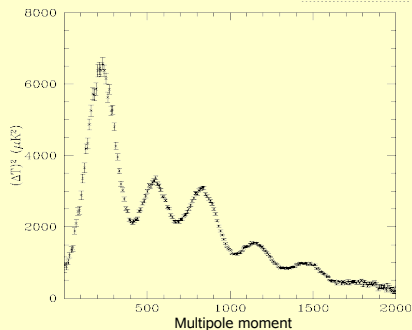


## From science to technology

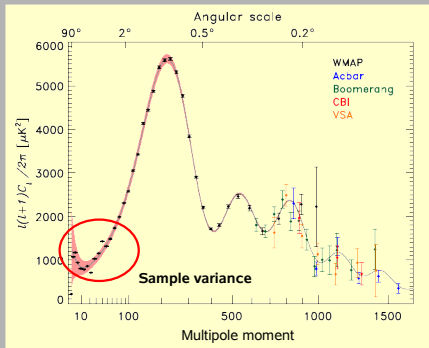
WMAP



Planck



## Sources of uncertainties - sample variance



- ▶ At low multipoles only a limited number of samples are available for  $C_\ell$  estimation

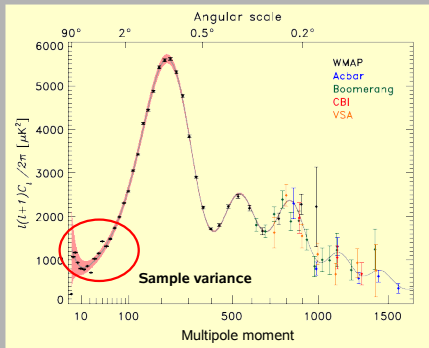
$$\delta C_\ell = \sqrt{\frac{2}{(2\ell + 1)}} C_\ell$$



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## Sources of uncertainties - sample variance

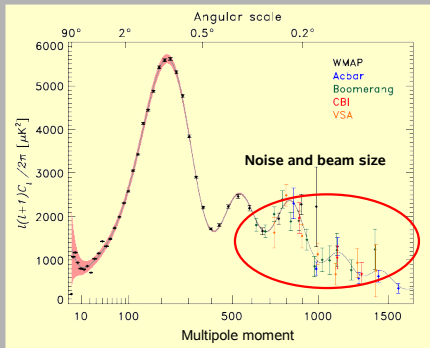


- ▶ At low multipoles only a limited number of samples are available for  $C_\ell$  estimation
- ▶ If only a fraction of the sky is observed then the uncertainty increases

$$\delta C_\ell = \sqrt{\frac{2}{(2\ell + 1)f_{\text{sky}}}} C_\ell$$



## Sources of uncertainties - noise and beam size

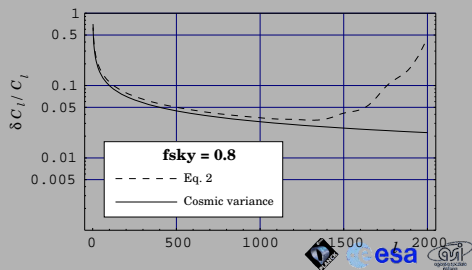
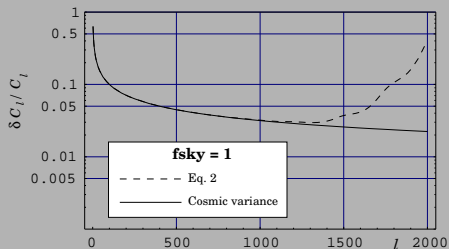
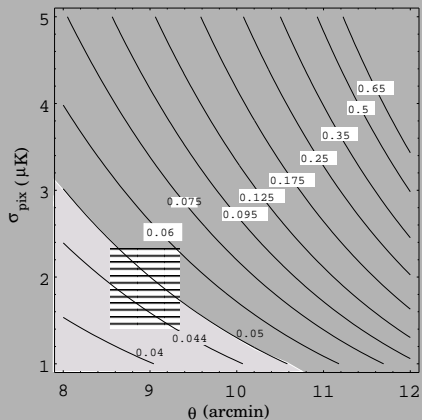


- ▶ At high multipoles the accuracy is limited mainly by noise and beam size

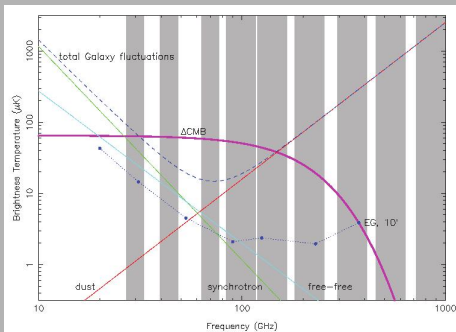
$$\delta C_\ell = \sqrt{\frac{2}{(2\ell + 1)f_{\text{sky}}}} \left( C_\ell + \frac{A\sigma_{\text{pix}}^2}{N_{\text{pix}}W_\ell^2} \right)$$

with  $W_\ell^2 = \exp[-\ell(\ell + 1)\sigma_B^2]$  and  
 $\sigma_B = \theta_{\text{FWHM}} / \sqrt{8 \ln(2)}$

# Requisites for Planck

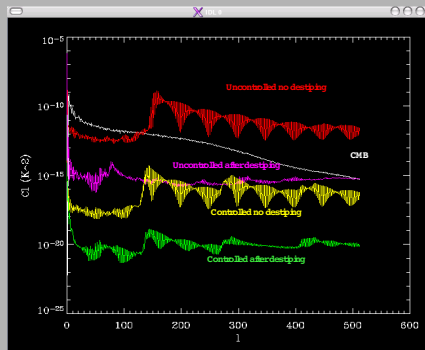


# Other sources of uncertainty



Optimal foreground subtraction with wide range of frequencies spanning more than 1 order of magnitude in frequency

# Other sources of uncertainty



Systematic effects must be controlled at the level of  $\sim 3\text{-}4 \mu\text{K}$  on the final sky pixel

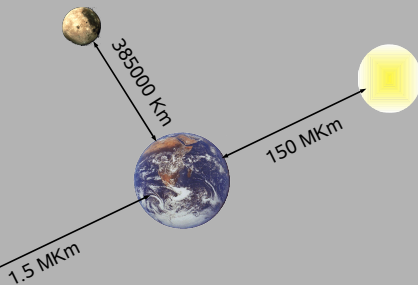
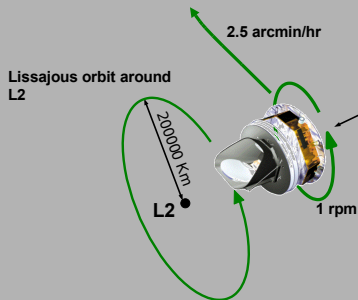


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# Orbit and scanning strategy

- Orbit Around L2
- 1 r.p.m. spin
- 2.5'/hr repointing
- Anti-solar spin axis
- Field of view at  $85^\circ$  with respect to spin axis
- 2 full sky surveys

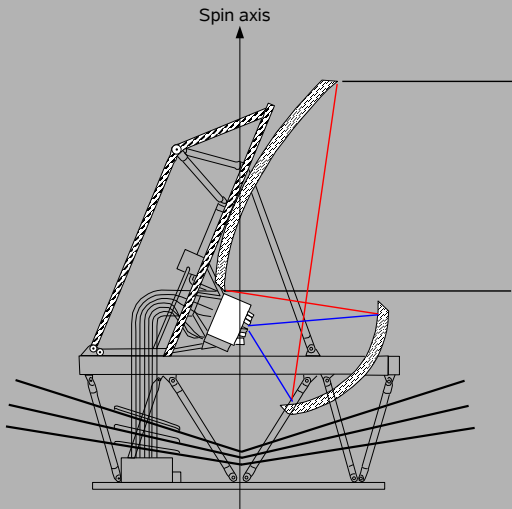


## ADVANTAGES

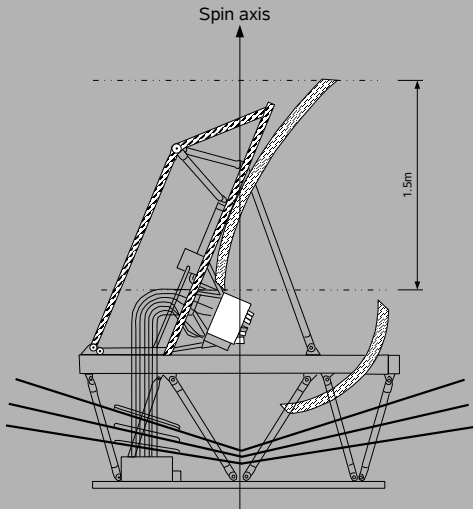
- Full sky view
- Optimal thermal stability
- Minimum straylight from Sun-Earth-Moon



# The Planck Telescope



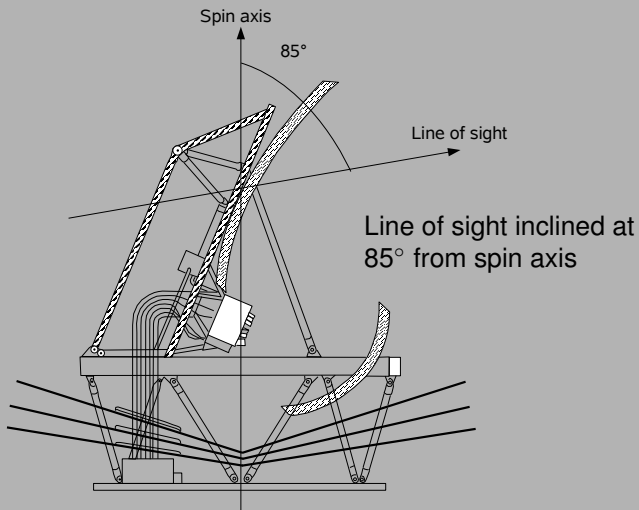
# The Planck Telescope



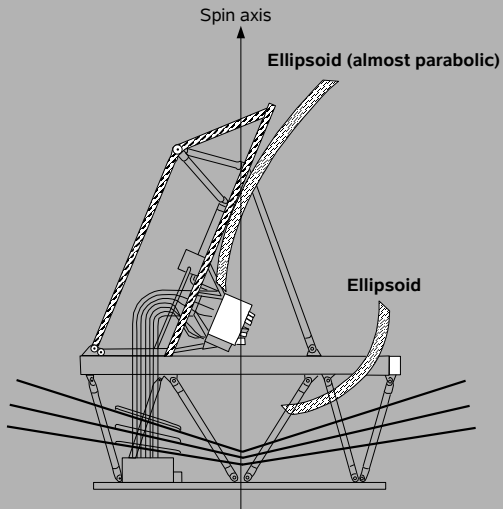
Primary aperture of 1.5 m  
provides  $\sim 9$  arcmin  
angular resolution at 100  
GHz



# The Planck Telescope



# The Planck Telescope



Aplanatic design  
minimises beam  
aberrations



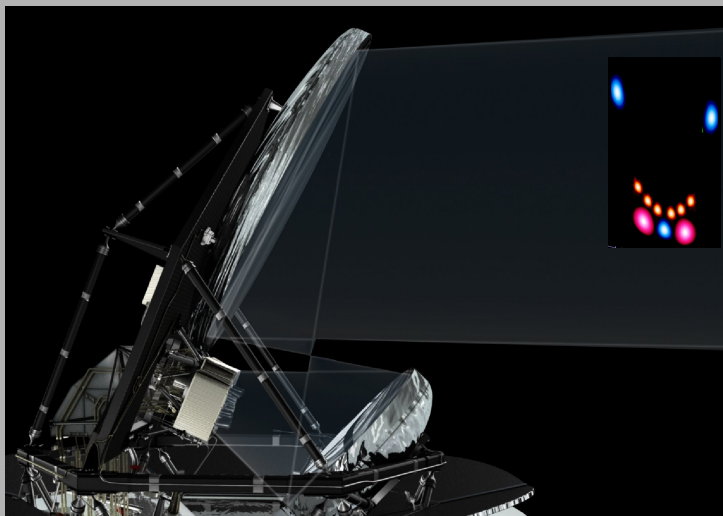
# The Planck Telescope



- ▶ Telescope realised in CFRP (Carbon Fiber Reinforced Plastic) for optimal weight and best thermal performances
- ▶ Hexagonal cell structure
- ▶ Surface made of aluminum coated by silicon oxide
- ▶ Emissivity  $< 1\%$

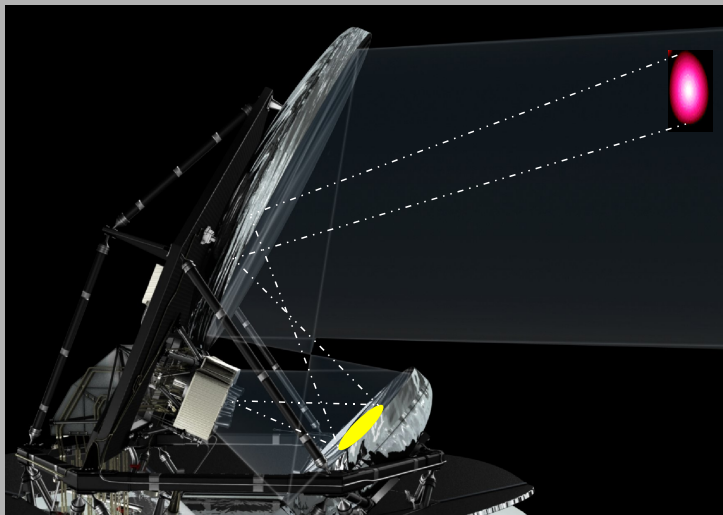


# Telescope-feed coupling



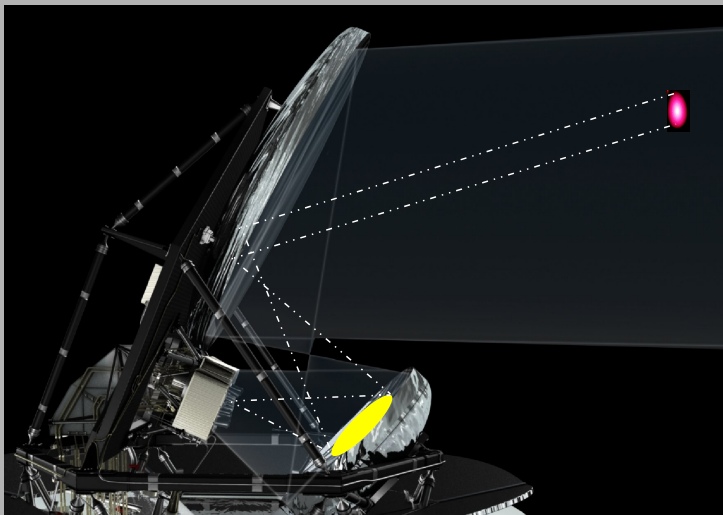
# Telescope-feed coupling

Underillumination  $\rightarrow$  poor angular resolution



# Telescope-feed coupling

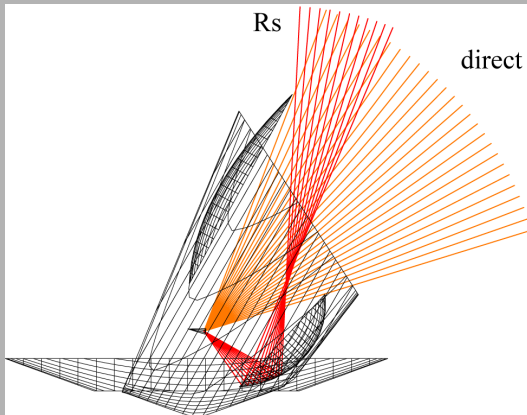
Overillumination  $\rightarrow$  improved resolution but...



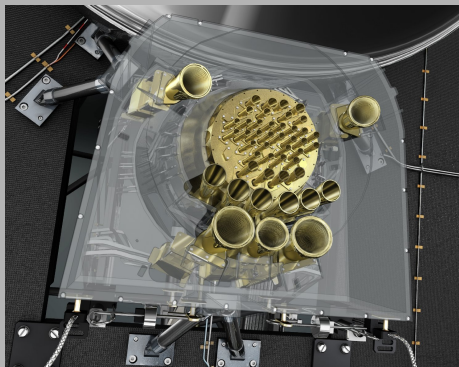


# Telescope-feed coupling

Overillumination  $\rightarrow$  improved resolution but... sensitivity to straylight



# Feed-horns [R.Nesti (LFI), B. Maffei (HFI)]

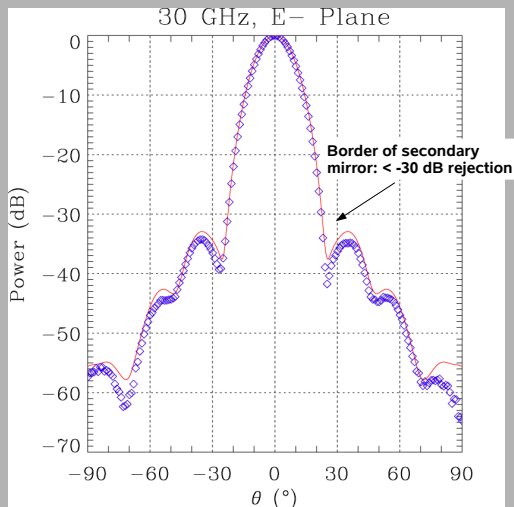
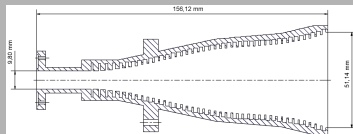


Dual profiled corrugated horns have been selected as the best solution in terms of:

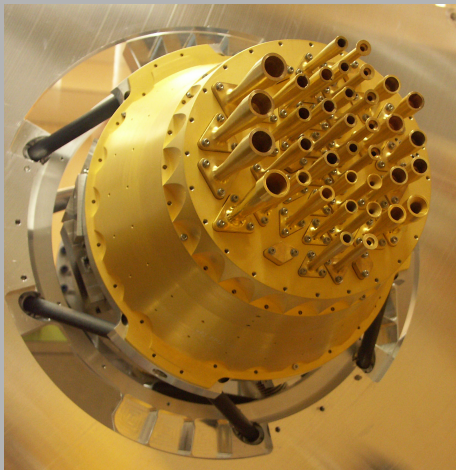
- ▶ level of cross polarization
- ▶ level of sidelobes
- ▶ return and insertion losses
- ▶ shape of the main lobe
- ▶ low weight
- ▶ compactness



# Feed-horns [R.Nesti (LFI), B. Maffei (HFI)]



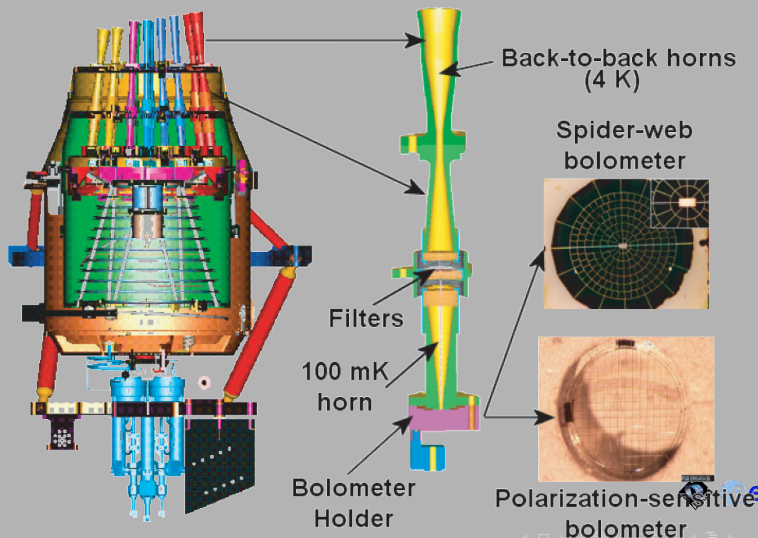
# High Frequency Instrument



- ▶ Array of 52 bolometric detectors cooled at 0.1 K in 6 frequency bands between 100 and 857 GHz
- ▶ Focal plane feed horns cooled at 4 K
- ▶ Sensitive to polarisation at four lower frequencies
- ▶ Developed by consortium lead by IAS-Orsay (P.I. J-L. Puget, I.S. J-M. Lamarre)



# High Frequency Instrument - details



# High Frequency Instrument - performances

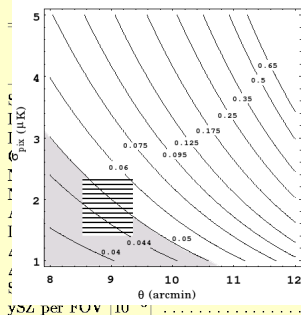
HFI PERFORMANCE GOALS<sup>a</sup>

INSTRUMENT CHARACTERISTIC	CENTER FREQUENCY [GHz]					
	100	143	217	353	545	857
Spectral resolution $\nu/\Delta\nu$ .....	3	3	3	3	3	3
Detector technology .....	Spider-web and polarisation-sensitive bolometers					
Detector temperature .....	0.1 K					
Cooling system .....	20 K Sorption Cooler + 4 K J-T + 0.1 K Dilution					
Number of spider-web bolometers .....	0	4	4	4	4	4
Number of polarisation-sensitive bolometers .....	8	8	8	8	0	0
Angular resolution [FWHM arcminutes] .....	9.5	7.1	5.0	5.0	5.0	5.0
Detector Noise-Equivalent Temperature [ $\mu\text{Ks}^{0.5}$ ] .....	50	62	91	277	1998	91000
$\Delta T/T$ Intensity <sup>b</sup> [ $10^{-6}\mu\text{K/K}$ ] .....	2.5	2.2	4.8	14.7	147	6700
$\Delta T/T$ Polarisation (U and Q) <sup>b</sup> [ $10^{-6}\mu\text{K/K}$ ] .....	4.0	4.2	9.8	29.8	...	...
Sensitivity to unresolved sources [mJy] .....	12.0	10.2	14.3	27	43	49
ySZ per FOV [ $10^{-6}$ ] .....	1.6	2.1	615	6.5	26	605

<sup>a</sup> Goal sensitivities. All subsystems have been designed to reach or exceed the performances of this table, which are expected to be achieved in orbit. Sensitivity requirements are a factor of two worse, and would still achieve the core scientific objectives of the mission.

<sup>b</sup> Average  $1\sigma$  sensitivity per pixel (a square whose side is the FWHM extent of the beam), in thermodynamic temperature units, achievable after 2 full sky surveys (14 months).

# High Frequency Instrument - performances



FI PERFORMANCE GOALS<sup>a</sup>

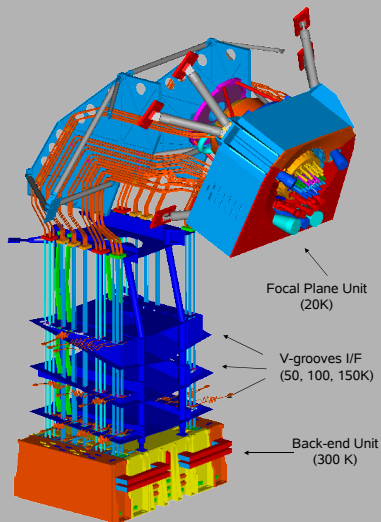
C	CENTER FREQUENCY [GHz]					
	100	143	217	353	545	857
.....	3	3	3	3	3	3
.....	Spider-web and polarisation-sensitive bolometers					
.....	0.1 K					
.....	20 K Sorption Cooler + 4 K J-T + 0.1 K Dilution					
.....	0	4	4	4	4	4
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# Low Frequency Instrument

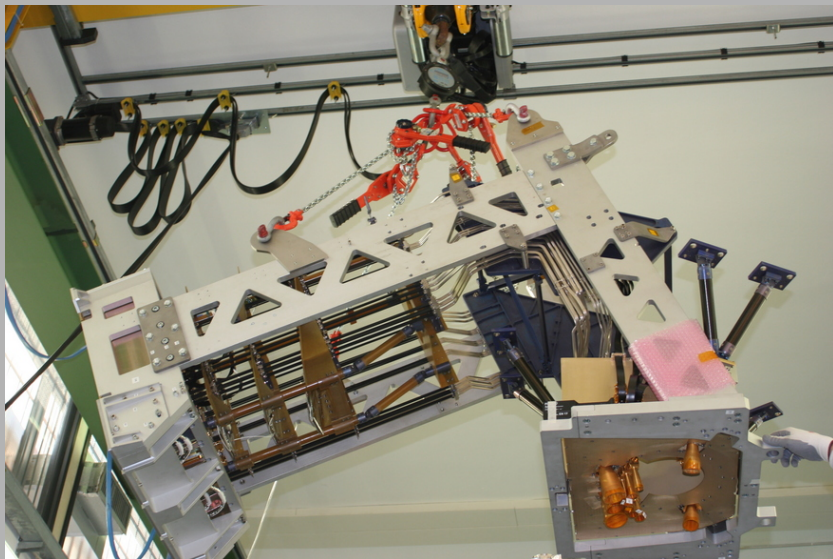


- ▶ Array of 44 radiometric detectors cooled at 20 K in 3 frequency bands centred at 30, 44 and 70 GHz
- ▶ Sensitive to polarisation at all frequencies
- ▶ Developed by consortium lead by INAF-IASF (P.I. R. Mandolesi, I.S. M. Bersanelli)
- ▶ Thales Alenia Space (formerly Laben) - Milan as industrial partner





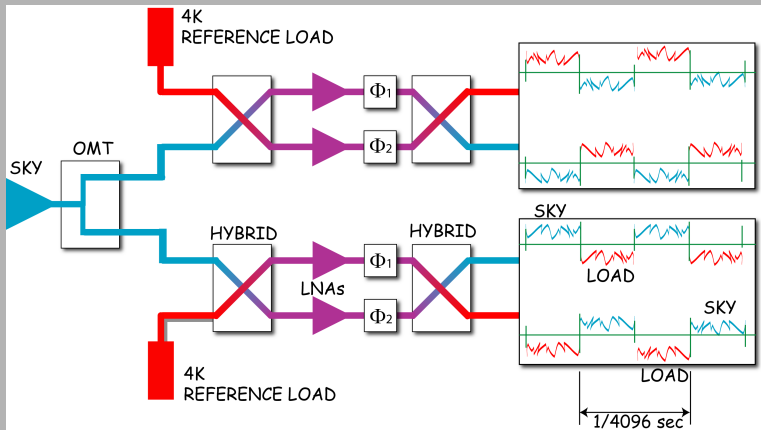
# Low Frequency Instrument - flight hardware



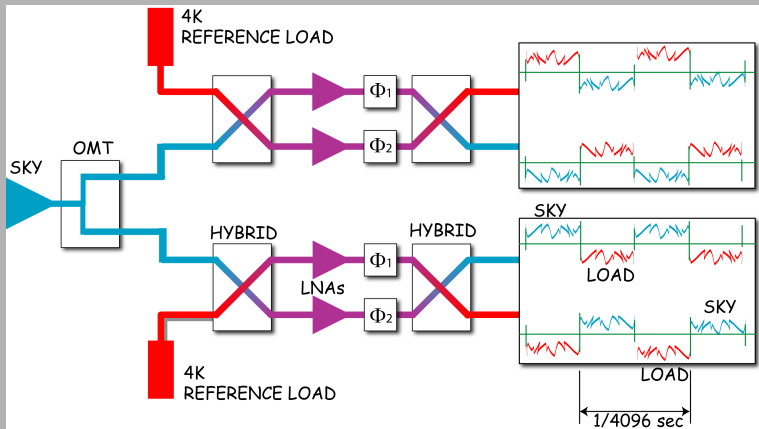
# Low Frequency Instrument - flight hardware



# The Planck-LFI pseudo-correlation receivers

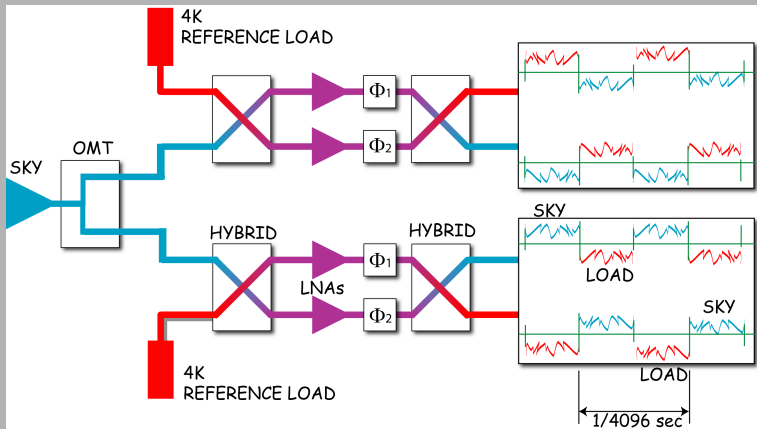


# The Planck-LFI pseudo-correlation receivers



Four outputs: R0D0, R0D1, R1D0, R1D1

# The Planck-LFI pseudo-correlation receivers



$$\Delta V(t) = V_{sky}(t) - r \times V_{load}(t)$$

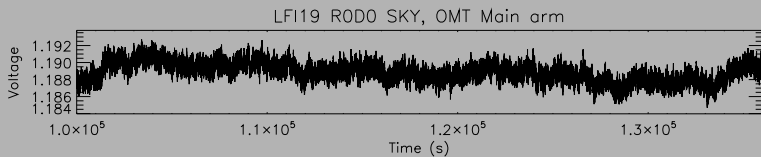
$$r \sim \langle V_{sky} \rangle / \langle V_{load} \rangle$$



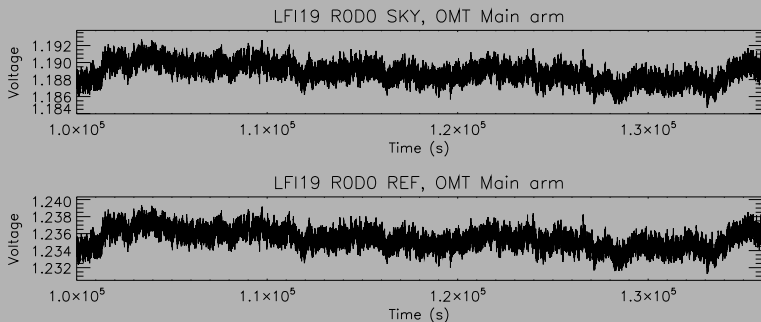
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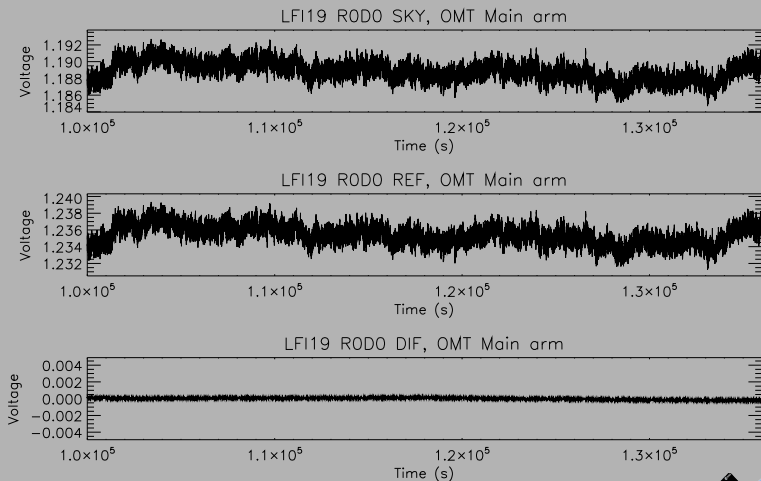
# The Planck-LFI pseudo-correlation receivers



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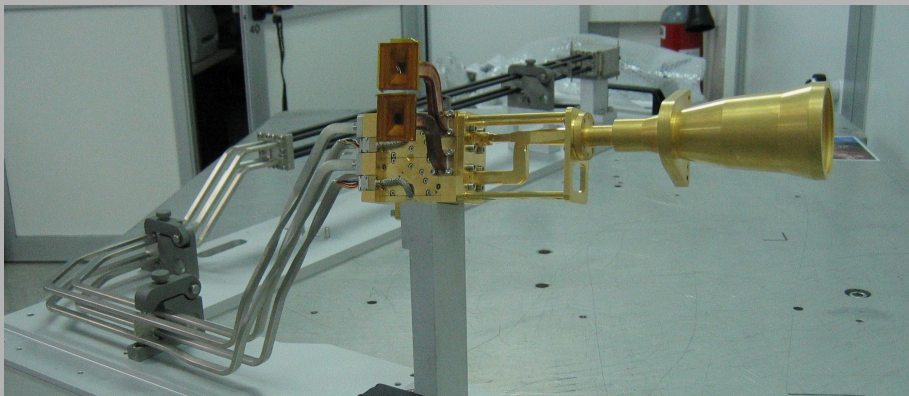


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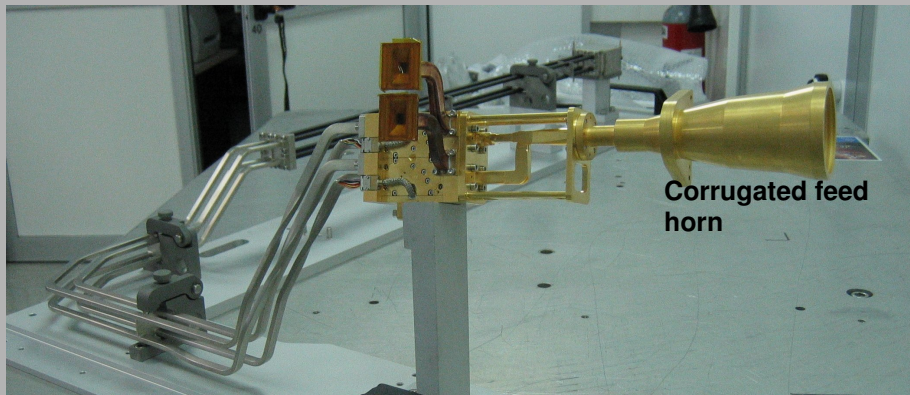




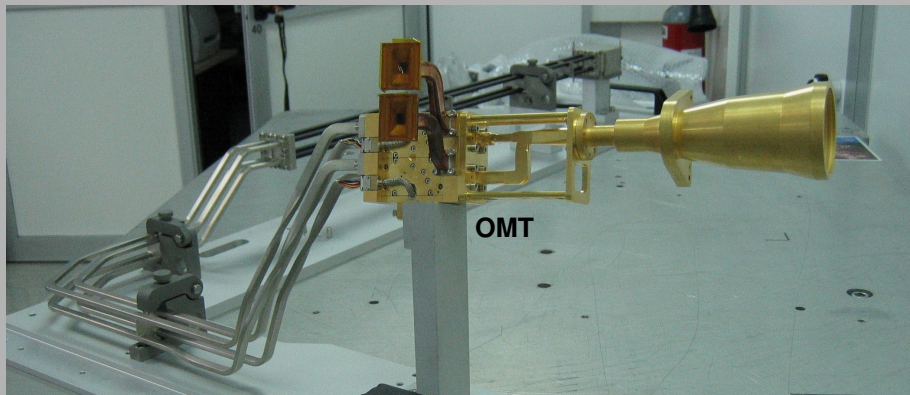
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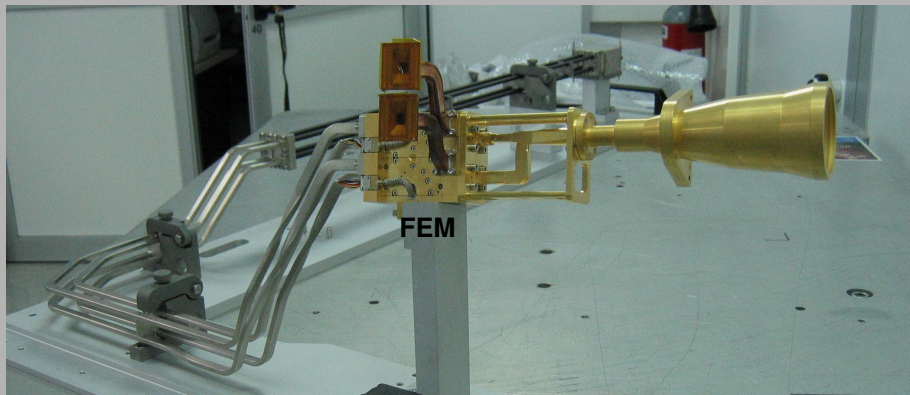
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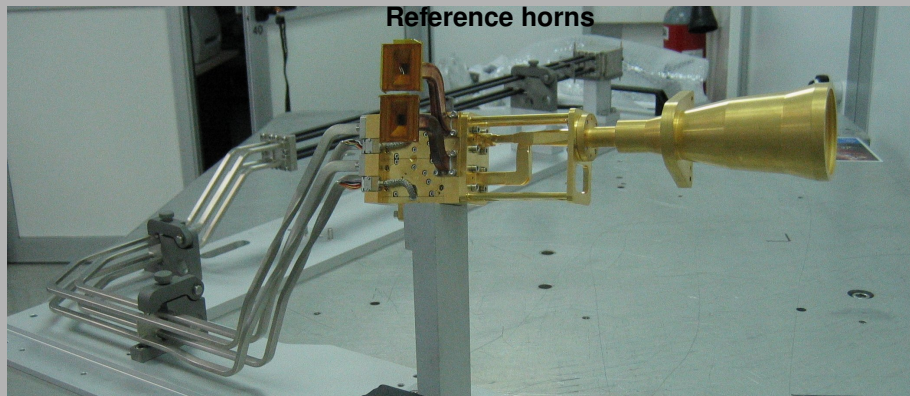
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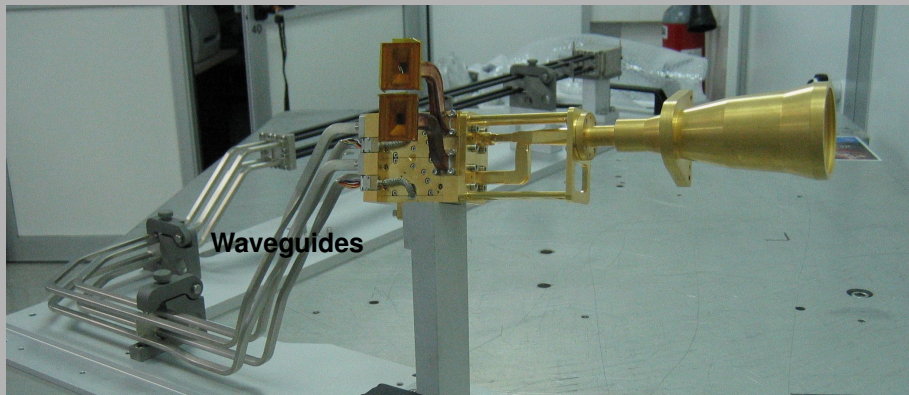
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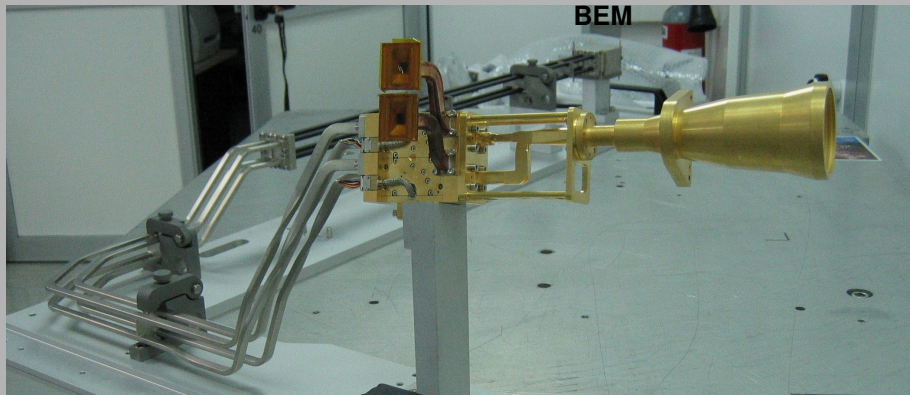
# The Planck-LFI pseudo-correlation receivers



# The Planck-LFI pseudo-correlation receivers



# The Planck-LFI pseudo-correlation receivers



# Low Frequency Instrument - performances

LFI PERFORMANCE GOALS<sup>a</sup>

INSTRUMENT CHARACTERISTIC	CENTER FREQUENCY [GHz]		
	30	44	70
InP HEMT Detector technology	MIC		MMIC
Detector temperature	20 K		
Cooling system	H <sub>2</sub> Sorption Cooler		
Number of feeds	2	3	6
Angular resolution [arcminutes FWHM]	33	24	14
Effective bandwidth [GHz]	6	8.8	14
Sensitivity [mK Hz <sup>-1/2</sup> ]	0.17	0.20	0.27
System temperature [K]	7.5	12	21.5
Noise per 30' reference pixel [ $\mu$ K]	6	6	6
$\Delta T/T$ Intensity <sup>b</sup> [ $10^{-6} \mu$ K/K]	2.0	2.7	4.7
( $\Delta T/T$ ) Polarisation (Q and U) <sup>b</sup> [ $\mu$ K/K]	2.8	3.9	6.7
Maximum systematic error per pixel [ $\mu$ K]	< 3	< 3	< 3

<sup>a</sup> All subsystems are designed to reach or exceed the performances of this table.

<sup>b</sup> Average  $1\sigma$  sensitivity per pixel (a square whose side is the FWHM extent of the beam), in thermodynamic temperature units, achievable after 2 full sky surveys (14 months).



# Cold is cool!

A **cold** and **stable** environment is key for telescope, focal plane and instruments:

- ▶ High sensitivity requirements call for cold optics and detectors
- ▶ Bolometric detectors need to be cooled at sub-K level to work (0.1 K for Planck-HFI)
- ▶ Stringent systematic effect control requires stable thermal conditions.



# Cold is cool!

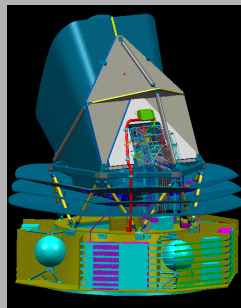
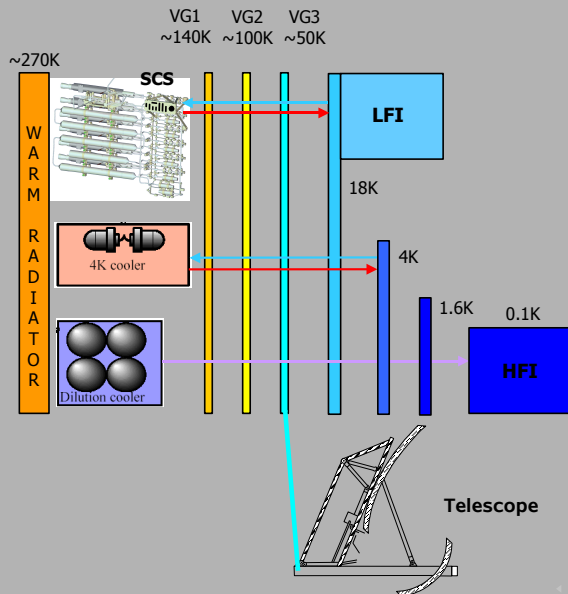
A **cold** and **stable** environment is key for telescope, focal plane and instruments:

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- ▶ Stringent systematic effect control requires stable thermal conditions.

**Planck is an extraordinary technological challenge from the thermal point of view**



# The Planck cryo-chain



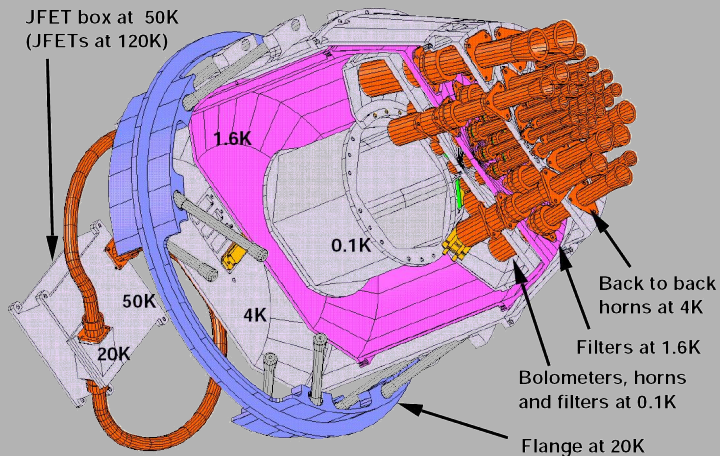
# Passive cooling



- ▶ About 16 m<sup>2</sup>
- ▶ Open Honeycomb, black painted
- ▶ Reflection of thermal IR to space
- ▶ Cryocoolers heat exchangers on V-grooves
- ▶ Cryo structure support by fiberglass struts
- ▶ Parasitics interception of harness, struts, waveguides

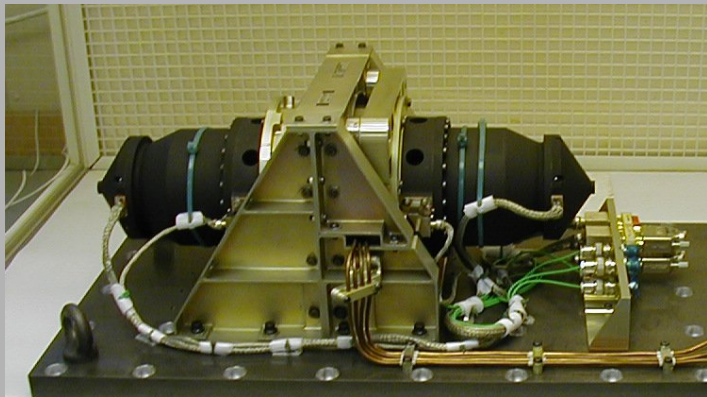


# The HFI cold stages (4K, 1.6K and 0.1K)



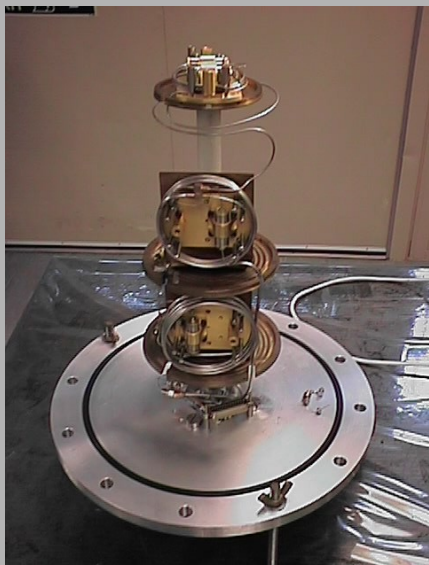
# The 4 K mechanical cooler [RAL - UK]

## Mechanical compressor



# The 4 K mechanical cooler [RAL - UK]

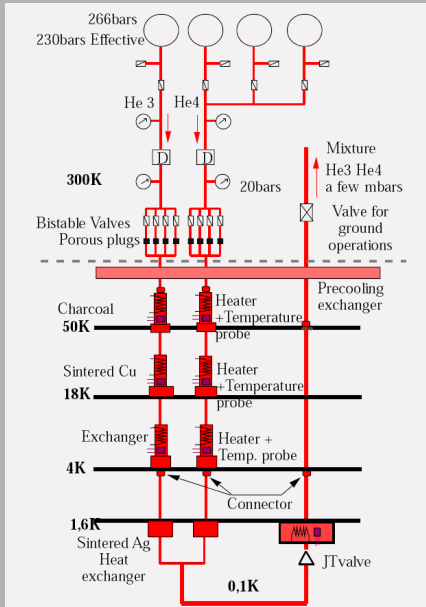
## J-T cold end



- ▶ Vibrations active control
- ▶ Heatlift 14mW @ 4.5K
- ▶ Input power 60W
- ▶ Total mass 40kg



# The 0.1 K dilution cooler



- ▶ Capillary dilution, open cycle
- ▶ Heatlift 100 nW at 100 mK
- ▶ Built by Air Liquide
- ▶ Precooling at 50K, 20K, 4K.
- ▶ J-T expansion at 1.6K
- ▶ 4 tanks of 51 l at 295b (1 for 3He, 3 for 4He)

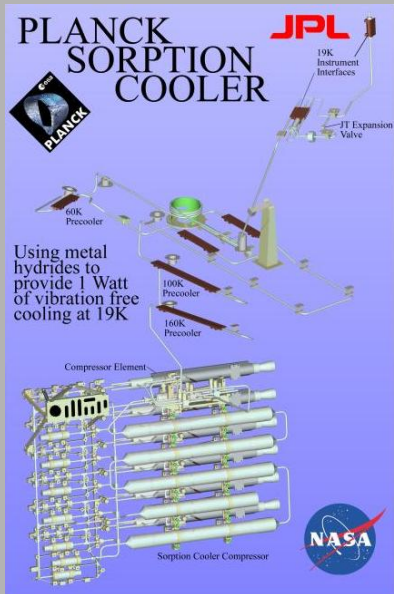


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# The Planck Sorption Cooler



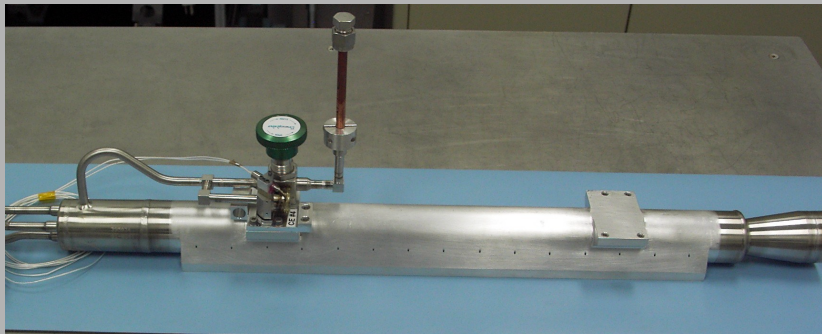
## JPL has a pioneering heritage on hydrogen sorption coolers:

- ▶ 10K prototype demonstrator in 1992 (single shot)
- ▶ BETSCE 10K solid H<sub>2</sub> tested on Space Shuttle in 1996 (single shot)
- ▶ Planck SCS first H<sub>2</sub> continuous cycle chemi-sorption cooler used in space
- ▶ Engineering Breadboard & FM's



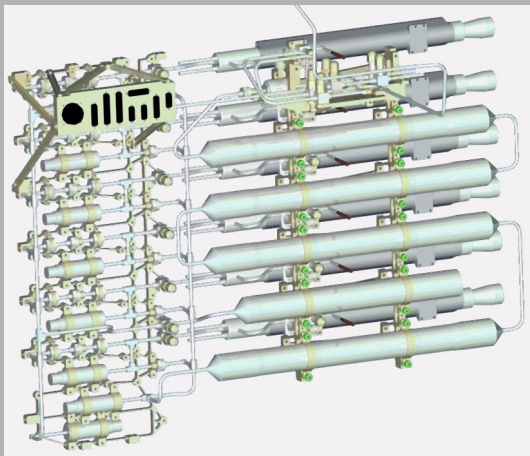
# How the Sorption Cooler works

Each compressor element is a tube containing a metal hydride that absorbs or releases hydrogen depending on its temperature

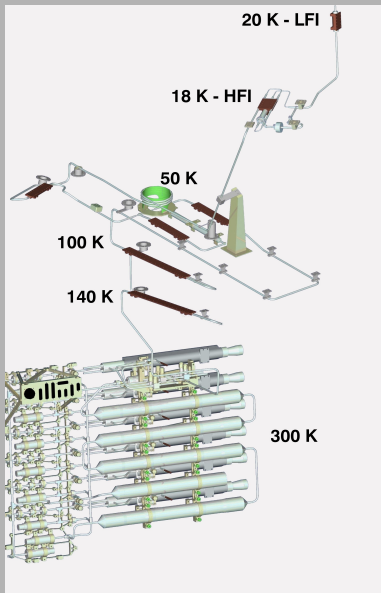


# How the Sorption Cooler works

Six compressors alternatatively warming up and cooling down generate a constant high pressure hydrogen flow



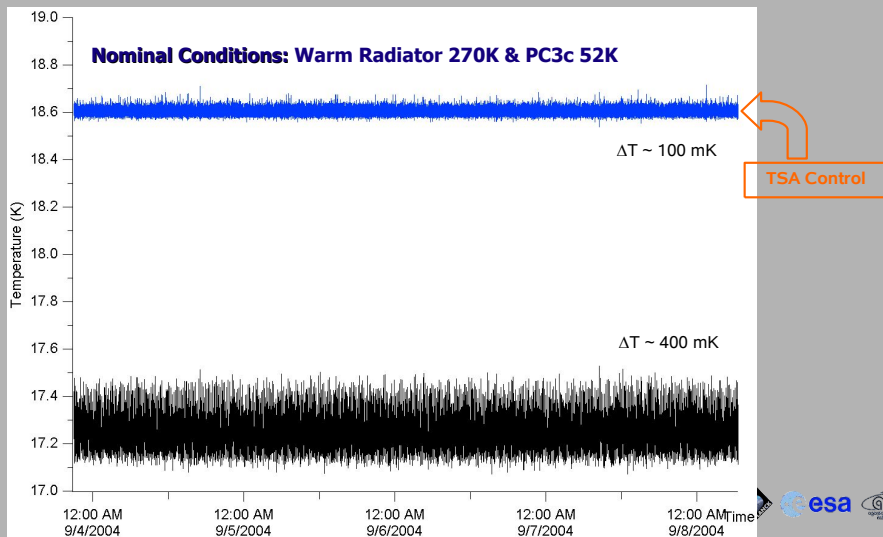
# How the Sorption Cooler works



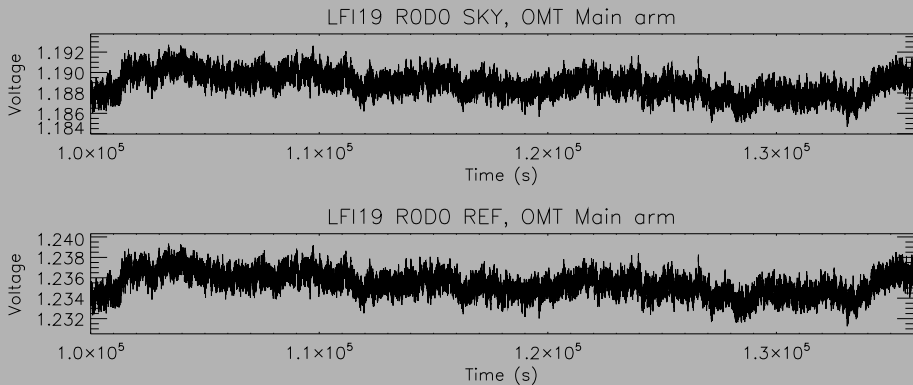
- ▶ Hydrogen flows in the Cooler piping from SVM to payload
- ▶ Three thermal interfaces with V-grooves precool the fluid
- ▶ Hydrogen expands and cool in two heat exchangers connected to HFI and LFI



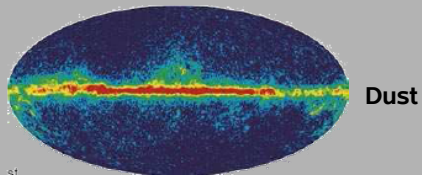
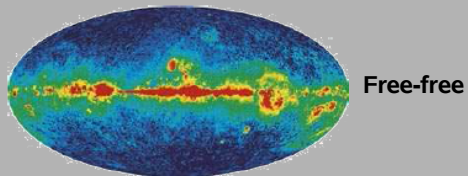
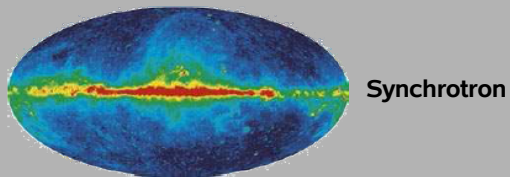
# Sorption cooler temperature stability



# Data processing - from time ordered data ...

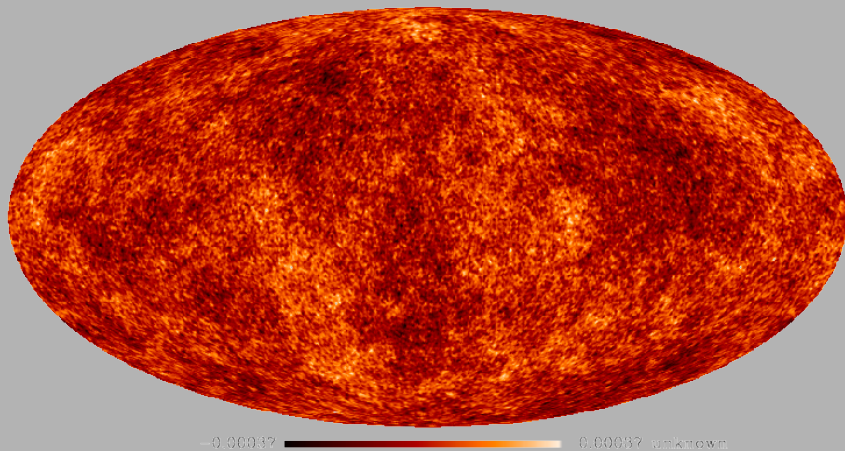


## Data processing - ... to components maps ...



s1

## Data processing - ... to cmb map ...

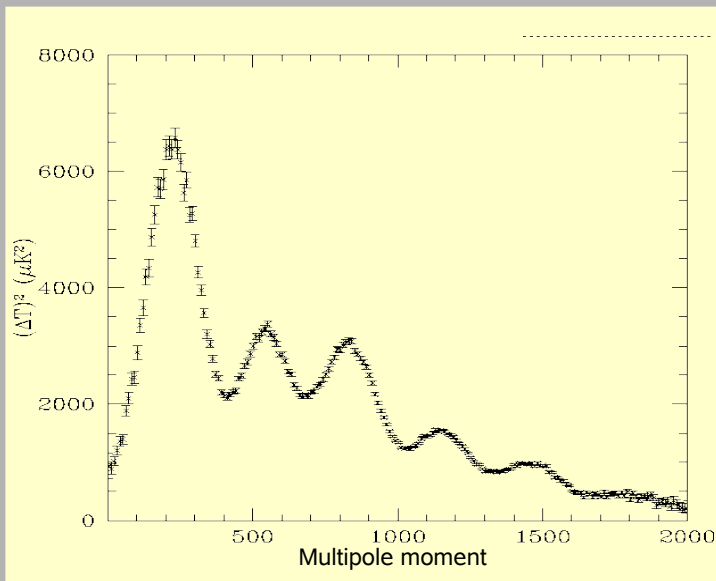


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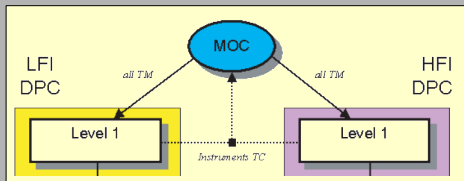




## Data processing - ... to power spectrum

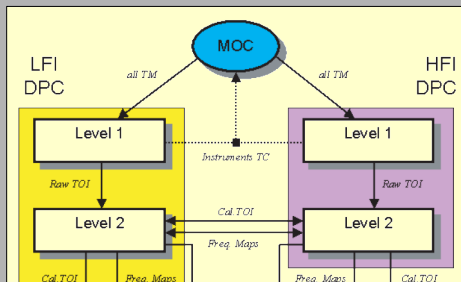


# Data processing structure



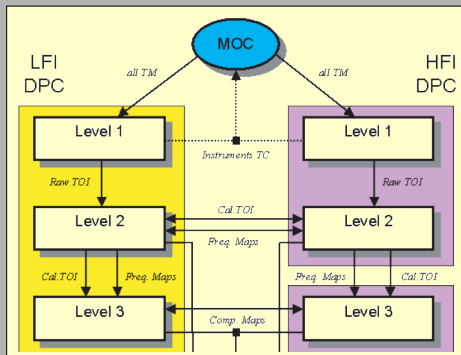
- ▶ Level 1: telemetry processing, generation of TOIs, instrument health checks

# Data processing structure



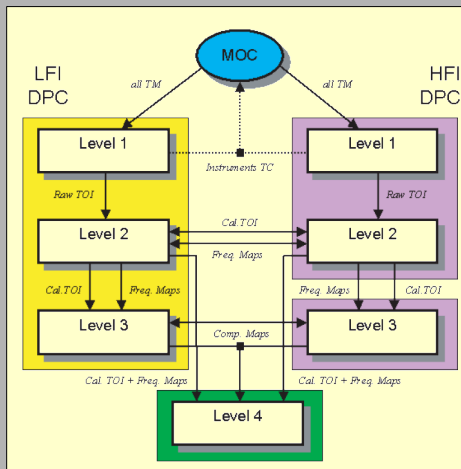
- ▶ Level 1: telemetry processing, generation of TOIs, instrument health checks
- ▶ Level 2: TOI processing, calibration and systematic error removal

# Data processing structure



- ▶ Level 1: telemetry processing, generation of TOIs, instrument health checks
- ▶ Level 2: TOI processing, calibration and systematic error removal
- ▶ Level 3: Map generation at various frequencies, component separation

# Data processing structure



- ▶ Level 1: telemetry processing, generation of TOIs, instrument health checks
- ▶ Level 2: TOI processing, calibration and systematic error removal
- ▶ Level 3: Map generation at various frequencies, component separation
- ▶ Level 4: Generation of final products



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# The Planck collaboration

